PIC Simulations 1 Tutorial lecture

<u>Yohei Miyake</u>^{1*} and Yoshiharu Omura² 1. Kobe University, Japan, 2. Kyoto University, Japan * y-miyake@eagle.kobe-u.ac.jp

> 15th International School for Space Simulations (ISSS-15), Aug 1, 2024, @Garching, Germany

Space plasma

4th state of matter

So hot: atoms split up into electrons & ions

Quasi-neutral: same numbers of e & i

Electrons & ions moving independently

Dynamics influenced by electromagnetic forces

Plasma current generates electromagnetic field

Exhibiting "collective" behaviors



Collective effects in shortest spatio-temporal scales > Debye shielding > Plasma oscillation



(electron) Plasma frequency:

$$\omega_{pe} = \sqrt{\frac{n_{\rm e}e^2}{m_{\rm e}\varepsilon_0}}$$

Behavior as "plasma" manifests on spatio-temporal scales larger than these characteristic quantities.

Criteria for plasmas

Criterion 1: System scales should be greater than λ_D in space and than ω_{pe}^{-1} in time.

Criterion 2: <u>Debye sphere should contain large number of particles.</u> ...coming from the condition that the potential energy by a nearest particle should be much smaller than the particle's kinetic energy ("weakly-coupled").



Particle-in-Cell (PIC) simulations



Design of "plasma behavior" in PIC code

Plasma criterion 2: Debye sphere must contain large number of particles. ...request us to solve huge particles

 \rightarrow <u>Computationally too expensive!</u>

Solution 1: use of a computational particle • that combine many real-world particles into one.

Caveat: a small number of particles with larger charge causes too large electrostatic interactions between the particles.

Solution 2: use of a "thick" particle or a "charge-cloud" particle, to reduce short-range inter-particle collisions. This is often referred to as a "super-particle".







Collisionless nature of super-particles



Numerical procedures of PIC simulations



Maxwell's equations $\nabla \times \boldsymbol{B} = \mu_0 \boldsymbol{J} + \frac{1}{c^2} \frac{\partial \boldsymbol{E}}{\partial t}$ $\nabla \times \boldsymbol{E} = -\frac{\partial \boldsymbol{B}}{\partial t}$ $\nabla \cdot \boldsymbol{E} = \frac{\rho}{\varepsilon_o}$ $\nabla \cdot \boldsymbol{B} = 0$ where $\varepsilon_0 \mu_0 = \frac{1}{c^2}$

Centered difference scheme

Define $u_i = u(xi)$. Then, Taylor-expand $u_{i+1/2}$ and $u_{i-1/2}$ around x_{i} .

$$u((i+1/2)\Delta x) = u_{i+1/2} = u_i + \frac{\Delta x}{2} \left(\frac{\partial u}{\partial x}\right)_i + \frac{1}{2!} \left(\frac{\Delta x}{2}\right)^2 \left(\frac{\partial^2 u}{\partial x^2}\right)_i + \frac{1}{3!} \left(\frac{\Delta x}{2}\right)^3 \left(\frac{\partial^3 u}{\partial x^3}\right)_i + \cdots$$

$$u((i-1/2)\Delta x) = u_{i-1/2} = u_i - \frac{\Delta x}{2} \left(\frac{\partial u}{\partial x}\right)_i + \frac{1}{2!} \left(\frac{\Delta x}{2}\right)^2 \left(\frac{\partial^2 u}{\partial x^2}\right)_i - \frac{1}{3!} \left(\frac{\Delta x}{2}\right)^3 \left(\frac{\partial^3 u}{\partial x^3}\right)_i + \cdots$$

where,
$$\Delta x = x_i - x_{i-1}$$
.

Subtraction of each other gives

$$u_{i+1/2} - u_{i-1/2} = \Delta x \left(\frac{\partial u}{\partial x}\right)_i + \frac{2}{3!} \left(\frac{\Delta x}{2}\right)^3 \left(\frac{\partial^3 u}{\partial x^3}\right)_i + \cdots$$

Centered difference expression

$$\left(\frac{\partial u}{\partial x}\right)_{i} = \frac{u_{i+1/2} - u_{i-1/2}}{\Delta x} + O(\Delta x^{2})$$
$$\sim \frac{u_{i+1/2} - u_{i-1/2}}{\Delta x}$$



Example of grid assignment (1D along x-axis)



Examples of grid assignment (2D & 3D)



Numerical procedures of PIC simulations



- Memory allocation
- Particle initialization
- Field initialization

Job completion Diagnostics





Current density in 1D system along *x*-axis

- J_y , J_z components: area-weighting method (same as charge density)
- J_x computed based on the area-weighting does NOT satisfy charge continuity equation: $\frac{d\rho}{dt} = -\nabla \cdot J = -\frac{\partial J_x}{\partial x}$...(1)
 - \rightarrow cause an accumulative error in an electrostatic (ES) field.
- Solution 1: Correcting an ES field every time step
 - 1. Define charge ρ_c associated with error in ES field : $\rho_c = \rho \nabla \cdot \mathbf{E}$
 - 2. Solve Poisson's equation: $\nabla^2 \phi_c = -\rho_c$, for ϕ_c : electric potential
 - 3. Compute ES field correction: $E_c = -\nabla \phi_c$
 - 4. Add E_c to E.
- Solution 2: Using a "Charge Conservation Method (CCM)" to compute current, which satisfies (1) in the machine accuracy.

Charge conservation method in 1D (J_x): case 1



Charge conservation method in 1D (J_x): case 2



Charge conservation methods in higher dimensions

Trajectory decomposition needed in case the particle moves across cell edges; various approaches proposed.



Numerical procedures of PIC simulations



Update of velocity: Buneman-Boris method



Relativistic equation of motion

$$\frac{\mathrm{d}}{\mathrm{d}t}(m\boldsymbol{v}) = q\left(\boldsymbol{E} + \boldsymbol{v} \times \boldsymbol{B}\right) \qquad \begin{array}{l} m = \gamma m_0 \\ \gamma = \frac{1}{\sqrt{1 - \left(\frac{v}{c}\right)^2}} \\ u = \frac{c}{\sqrt{c^2 - \left|\boldsymbol{v}\right|^2}} \boldsymbol{v} \qquad \boldsymbol{B}_u = \frac{c}{\sqrt{c^2 + \left|\boldsymbol{u}\right|^2}} \boldsymbol{B} \\ \frac{\mathrm{d}\boldsymbol{u}}{\mathrm{d}t} = \frac{q}{m_0} \left(\boldsymbol{E} + \boldsymbol{u} \times \boldsymbol{B}_{\boldsymbol{u}}\right) \end{array}$$

$$oldsymbol{v} = rac{c}{\sqrt{c^2 + \left|oldsymbol{u}
ight|^2}}oldsymbol{u}$$

Numerical procedures of PIC simulations



- Variable definition
- Memory allocation
- Particle initialization
- Field initialization

Job completion Diagnostics



Field interpolation to particle position



Electrostatic self-force cancellation

$$\rho_{i} = \frac{1}{\Delta x} \sum_{j}^{N_{p}} q_{j} \underline{W(x_{j} - X_{\overline{i}})} \\ \frac{\underline{E_{x,i+1/2} - E_{x,i-1/2}}}{\Delta x} = \frac{\rho_{i}}{\varepsilon_{0}}$$

$$E_{x}(x) = \sum_{i=1}^{N_{x}} E_{x,i+1/2} \underline{W(x - X_{\overline{i}+1/2})}$$
Self-force
Relocation
$$E_{x,i} = \frac{E_{x,i-1/2} + E_{x,i+1/2}}{2}$$

$$E_{x}(x) = \sum_{i=1}^{N_{x}} E_{x,i} \underline{W(x - X_{\overline{i}})}$$
No Self-force



Numerical procedures of PIC simulations



Time Step Chart



Effect of centered differentiation

$$E(X_i, t) = E_o \exp(ikX_i - i\omega t)$$

$$\frac{\partial E(X_i, t)}{\partial x} = \frac{E(X_i + \Delta x/2, t) - E(X_i - \Delta x/2, t)}{\Delta x}$$

$$= \frac{1}{\Delta x} [\exp(ik\Delta x/2) - \exp(-ik\Delta x/2)] E(X_i, t)$$

$$= i \frac{\sin(k\Delta x/2)}{\Delta x/2} E(X_i, t) = iKE(X_i, t)$$



Modified dispersion relation of light mode

Electromagnetic modes in vacuum

 $\omega^2 = c^2 k^2$

In centered difference scheme,

 $\Omega^2 = c^2 K^2$



Condition for numerical stability 1

Centered Difference Scheme in space and time

$$\Omega^2 = c^2 K^2$$
 $\Omega = \frac{\sin(\omega \Delta t/2)}{\Delta t/2}, \quad K = \frac{\sin(k \Delta x/2)}{\Delta x/2}$

For
$$k = rac{\pi}{\Delta x}$$
 we have $\sin(rac{\omega \Delta t}{2}) = rac{\Delta t}{\Delta x}c < 1$

Courant Condition

$$c\Delta t < \Delta x$$

Condition for numerical stability 2

Shortest wavelength scale to solved in the explicit PIC is

$$|k| \sim \frac{1}{\lambda_D}$$

where λ_D is the Debye length.

Replacing *k* with *K*

$$|\sin(k\Delta x/2))| \sim \frac{\Delta x}{2\lambda_D} < 1$$

Explicit PIC should satisfy the following stability condition

$$\Delta x < 2\lambda_D$$

Violation of the condition leads to numerical (unphysical) heating.

Numerical procedures of PIC simulations



Initial & boundary conditions

Initialization

- Particle loading
 - Positions:
 - Velocities: Maxwellian, shifted Maxwellian, loss-cone, ring, etc.
- Initial electrostatic field: Poisson's equation

Outer boundary condition

- Periodic boundaries
- Reflecting boundaries
- Open boundaries
 - Particle injection from outer edge
 - Non-reflective field boundary: masking method

Inner boundary condition

for object-plasma interaction study

Numerical solutions of Poisson's equation

 $\nabla \cdot E = \rho / \varepsilon_0$, $E = -\nabla \phi$ reduce to $\nabla^2 \phi = -\rho / \varepsilon_0$ (Poisson's equation). In 1D,

$$\frac{\partial^2 \phi}{\partial x^2} = -\frac{\rho}{\varepsilon_0}$$

In Fourier space,

$$k^2\hat{\phi} = \frac{\hat{\rho}}{\varepsilon_0}$$

In centered difference scheme,

$$K^2 \hat{\phi}_k = \frac{\hat{\rho}_k}{\varepsilon_0}$$
, where $K = \frac{\sin(k\Delta x/2)}{\Delta x/2}$

FFT-based Poisson solver,



In periodic system,

$$K^2 \hat{\phi}_0 = \frac{\hat{\rho}_0}{\varepsilon_0} = 0$$
 for $k = 0 \longrightarrow \sum_i \frac{\rho_i}{N_x} = 0$ "Charge neutral"

Cancellation of uniform current

Consider periodic 1D system along x-axis:

$$\frac{\partial \boldsymbol{J}_u}{\partial t} = \frac{n_e e^2}{m_e} \boldsymbol{E}_u \qquad \quad \frac{\partial \boldsymbol{E}_u}{\partial t} = -\boldsymbol{J}_u$$

...provide a solution in the following form:

$$\boldsymbol{J}_{u} = \boldsymbol{J}_{o} \exp\left(i\omega_{pe}t\right) \qquad \boldsymbol{E}_{u} = \frac{\imath}{\omega_{pe}}\boldsymbol{J}_{o} \exp\left(i\omega_{pe}t\right)$$

The uniform component must be subtracted. (Periodic system should also be "<u>current neutral</u>".)

$$\boldsymbol{J}_u = rac{1}{N_x} \sum_{i=1}^{N_x} \boldsymbol{J}_i \qquad \qquad \boldsymbol{J}_{i,sub} = \boldsymbol{J}_i - \boldsymbol{J}_u$$

Initial & boundary conditions

Initialization

- Particle loading
 - Positions:
 - Velocities: Maxwellian, shifted Maxwellian, loss-cone, ring, etc.
- Initial electrostatic field: Poisson's equation

Outer boundary condition

- Periodic boundaries
- Reflecting boundaries
- Open boundaries
 - Particle injection from outer edge
 - Non-reflective field boundary: masking method

Inner boundary condition

for object-plasma interaction study

Inner boundary conditions for object-in-plasma

Solid objects in space

- Celestial bodies ...(1)
- Dust grains ...(2)
- Spacecraft / instrument ...(3) ...and so on.

Effects at object surface-

- Particle loss
- Particle emission
- Charge deposition
- Surface potential
- Conducting current
- EM scattering







Under certain conditions, features called "electron wings" can form around spacecraft, potentially introducing interference and artifacts into data collected by onboard instruments.

By @MarkZastrow @Fysikk_UniOslo



"Electron Wings" Can Interfere with Spacecraft Measurements - Eos Spacecraft sometimes produce a form of electrical self-interference as they zip through plasmas in space—a previously unreported effect that may be lurking in... & eos.org

Capacitance matrix



Control of on-grid potential/charge



Some applications...

- 1. <u>Known</u> target potentials on all grids on object surface: Solve $Q = C\Phi$. Application: electrodes with applied potential
- 2. Equi-potential over object surface, but its potential value <u>unknown</u>: Solve $Q = C\Phi$ with $\sum_i \Delta q_i = 0$. Application: conducting objects with floating potential (e.g., spacecraft)



PIC simulations on Supercomputers

Fundamental strategies of parallelization



(static) Domain decomposition

Particle decomposition

Load balancing consideration



Advanced strategies for parallelization



Dynamic domain decomposition



References from past ISSSs

Advanced Methods for Space Simulations (ISSS-7)

"One-dimensional Electromagnetic Particle Code: KEMPO1" Edited by H. Usui and Y. Omura (2007)

Computer Space Plasma Physics:

Simulation Technique and Software (ISSS-4)

"KEMPO1: Technical Guide to one-dimensional electromagnetic particle code" Edited by H. Matsumoto and Y. Omura (1993)

Computer Simulation of Space Plasmas (ISSS-1)

"Particle simulation of electromagnetic waves and its application to space plasmas" Edited by H. Matsumoto and T. Sato (1985)