

Magnetic Reconnection Plasma Thruster

Fatima Ebrahimi (**Princeton University/PPPL**)

2024 IPELS-16 Workshop

Gharching, August 7, 2024



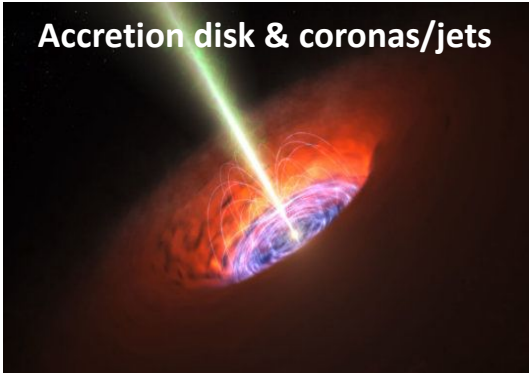
- **LDRD team**

**Art. Brook, Robert Lunsford, Nicholas O’Gorman,
Peter Titus, Jiawen Wang, Andrew Yard**

- **Philip Efthimion, Jon Menard, Roger Raman**

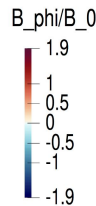
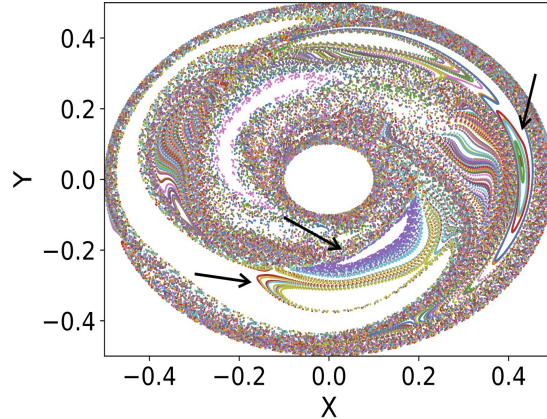


Accretion disk & coronas/jets



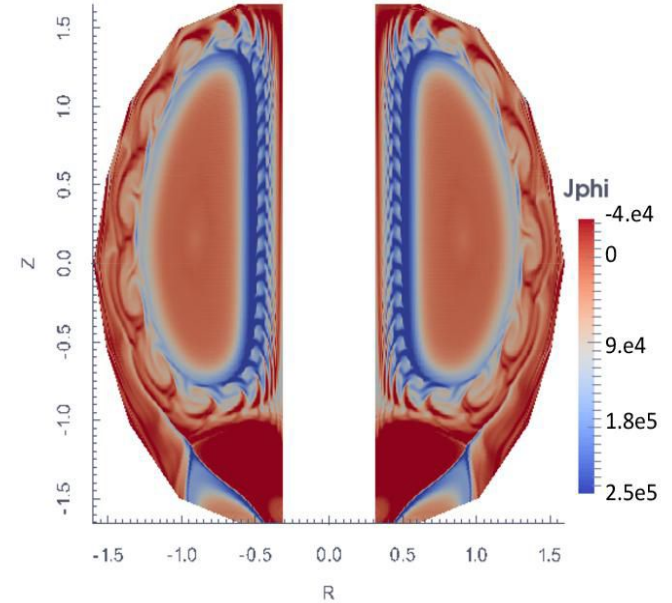
I -In astrophysical disks Magnetic islands in accretion flows

Toroidal Cross Section

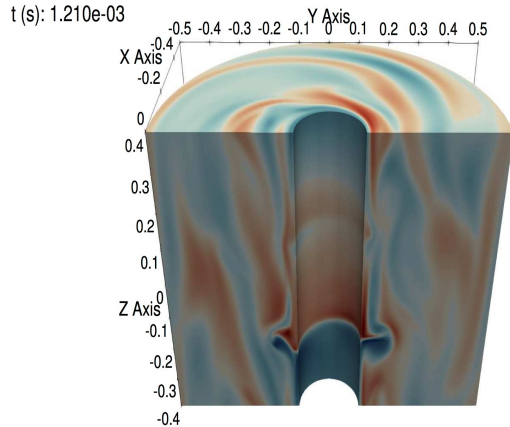


Rosenberg & Ebrahimi
ApJL 2021

II- In a tokamak ELM nonlinear dynamics



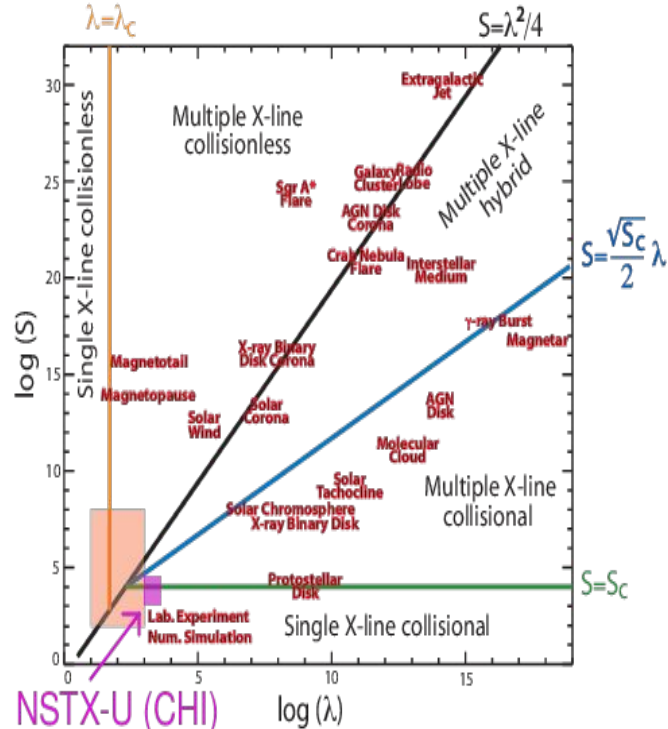
Onset and relaxation of SOL currents.
Ebrahimi PoP 2017,
Ebrahimi & Bhattacharjee NF 2023



From laboratory to large-scale fusion experiments (to explore multi-scale reconnection problem)

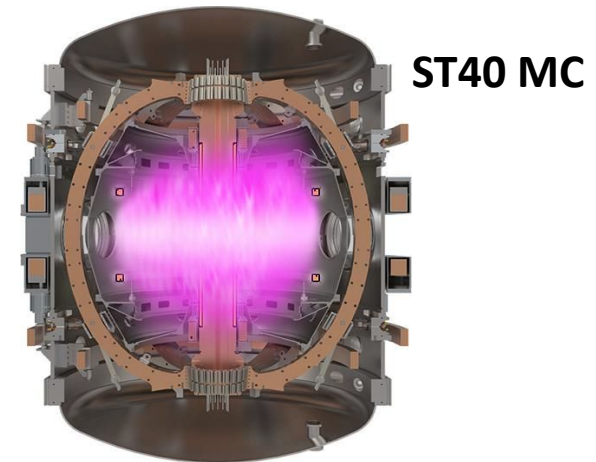
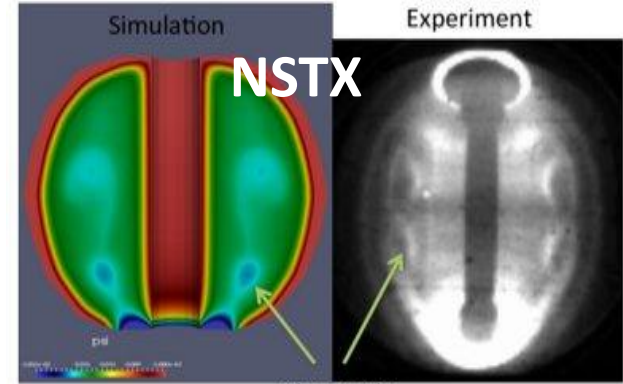
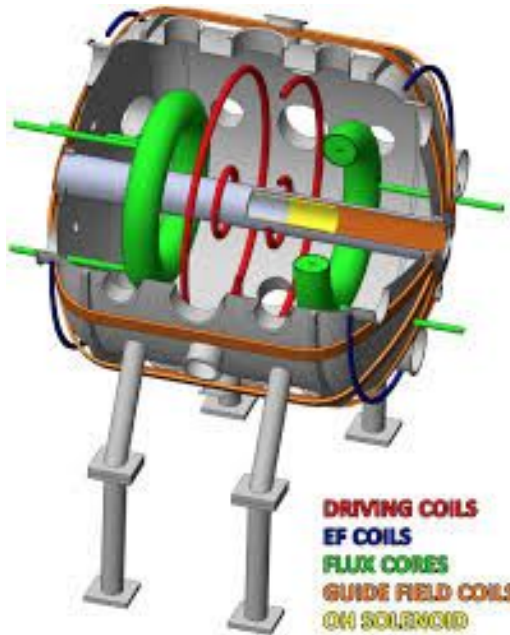


Reconnection phase diagram



Ji & Daughton 2011

Flare reconnection experiment (PPPL)

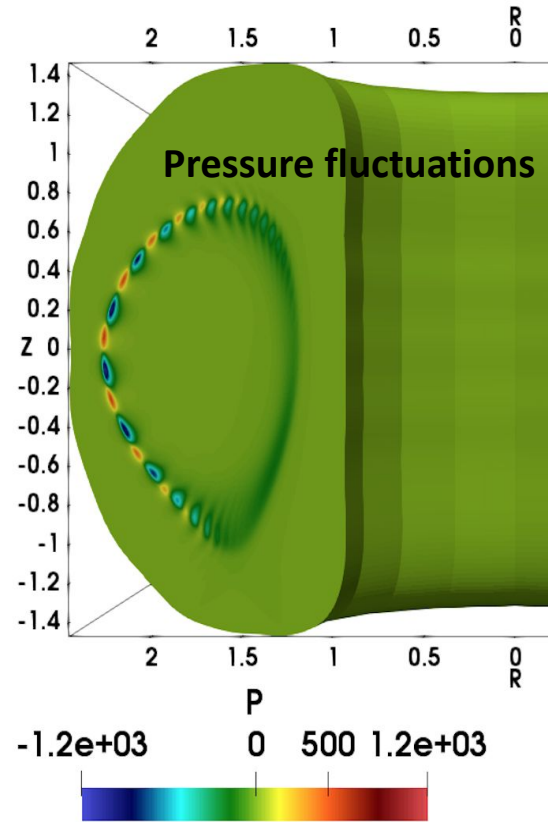
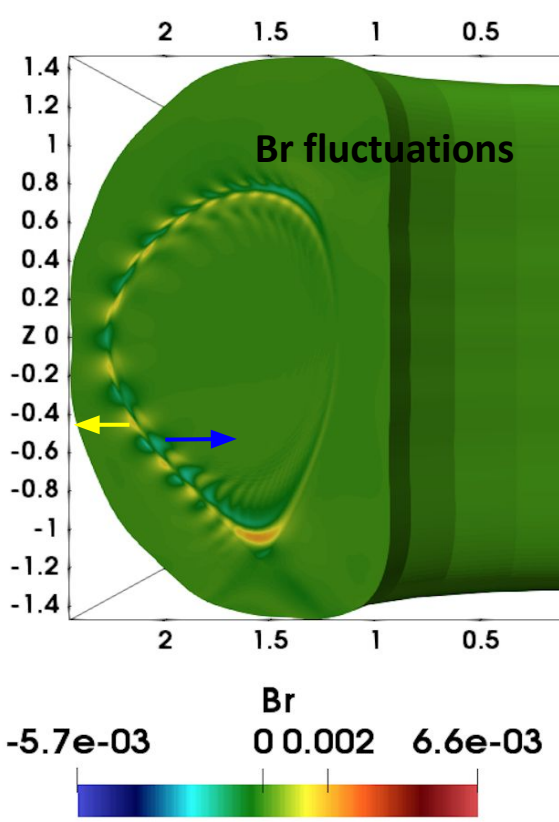


How could current sheets form?



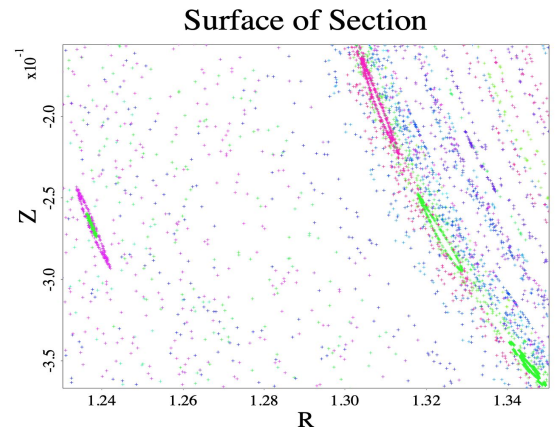
- 1- Due to a primary instability**
- 2- Nonlinear dynamics - turbulence (dynamical vortices)**
- 3- Externally forced reconnection to provide reconnection sites inductively or via helicity injection**
- 4- During flux expansion or due to strong current ramp up**

1) Nonlinear edge P-B modes do generate current sheets.

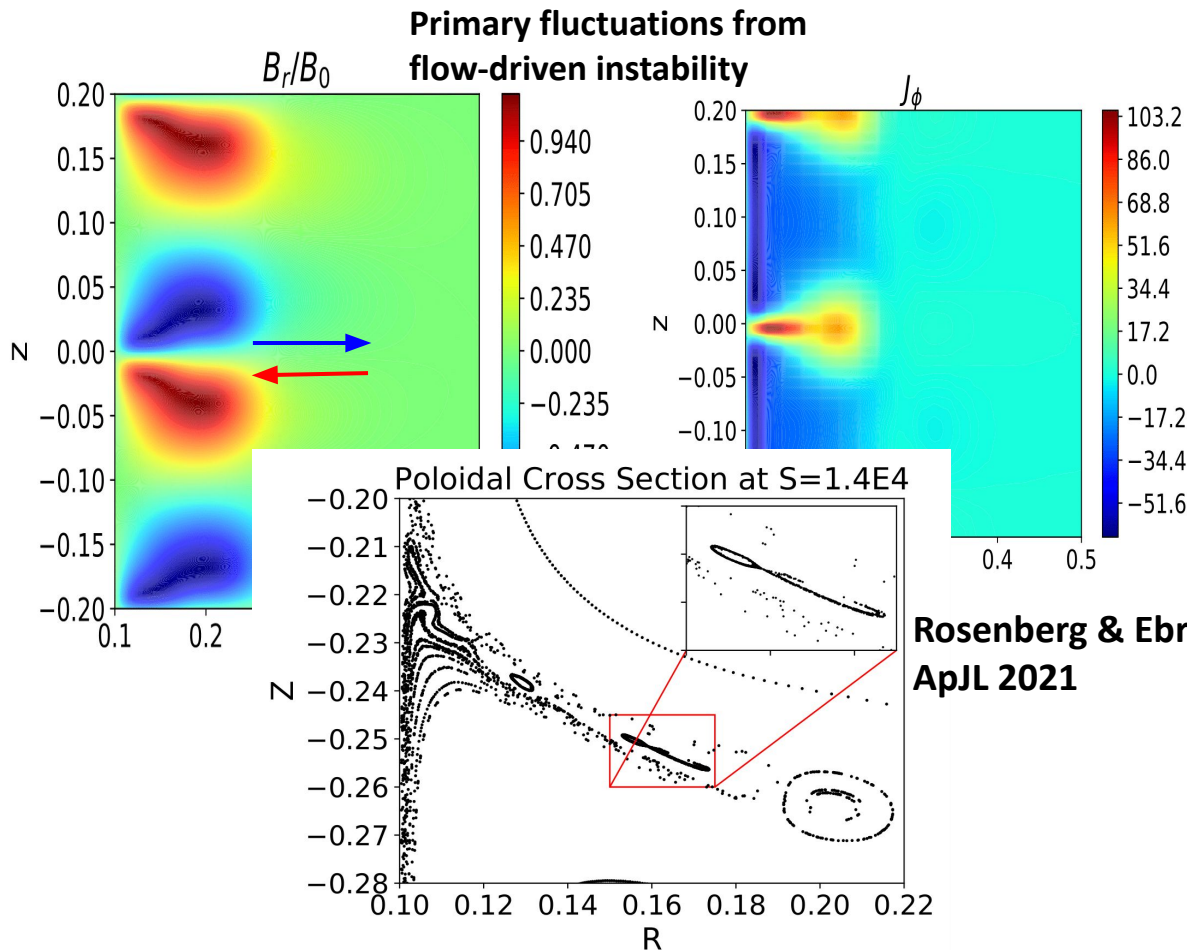
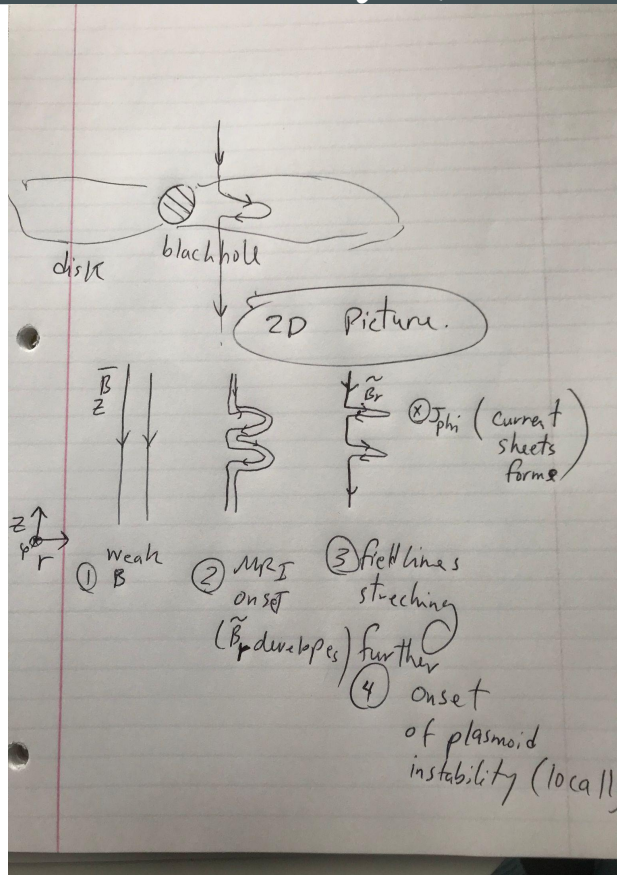


- As edge plasma interchangeably is displaced by the ballooning modes, local edge current sheet could form.

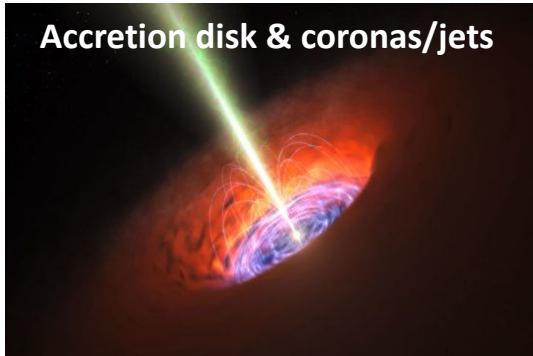
ELM crash: Reconnection physics via full extended MHD is needed for correct relaxation dynamics.



1) How about current sheet formation due to a primary instability? (and cause secondary reconnection)

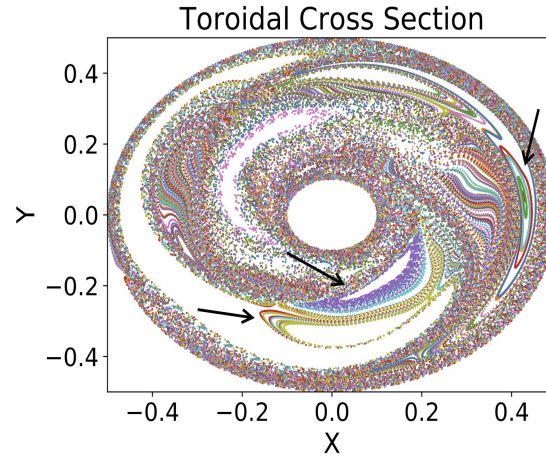


2- Magnetic reconnection sites are formed via 3D turbulence (dynamical vortices)



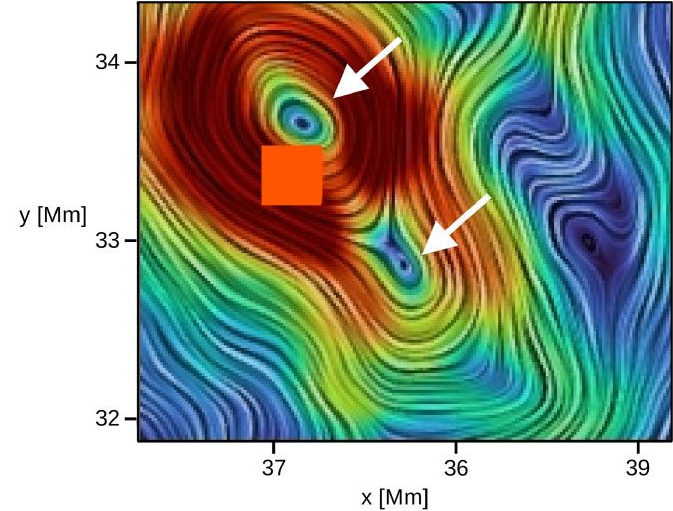
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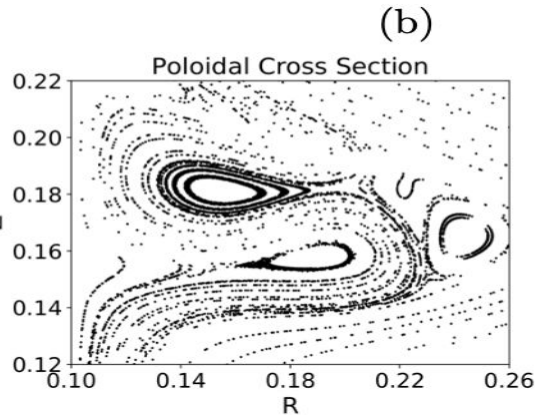


Rosenberg & Ebrahimi
ApJL 2021
Ebrahimi & Pharr ApJ 2022

II- Observation: Surface of the sun



Chian et al. MNRAS 2023



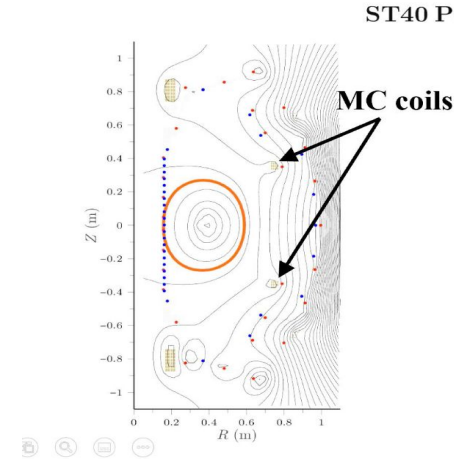
Merging current-sheets in 3D form plasmoids

3- Externally forced methods would provide reconnection sites



Compact ST-based fusion reactors require innovative plasma startup

- Merging compression on START, MAST, UTST, and recent ST40 experiments
- Transient coaxial helicity injection on HIT-II and NSTX
(This talk)
- Sustained and local helicity injection on HIT-II and Pegasus



Fundamental reconnection physics, including plasmoid physics, has been demonstrated during plasma start-up in NSTX



Fundamental reconnection physics, including plasmoid physics, has been demonstrated during plasma start-up in NSTX

PRL 114, 205003 (2015)

PHYSICAL REVIEW LETTERS

week ending
22 MAY 2015

Plasmoids Formation During Simulations of Coaxial Helicity Injection in the National Spherical Torus Experiment

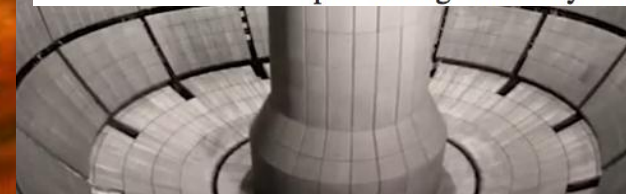
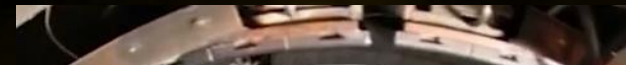
F. Ebrahimi¹ and R. Raman²

¹*Department of Astrophysical Sciences, Princeton University, Princeton, New Jersey 08543, USA*

²*University of Washington, Seattle, Washington 98195, USA*

(Received 2 March 2015; published 20 May 2015)

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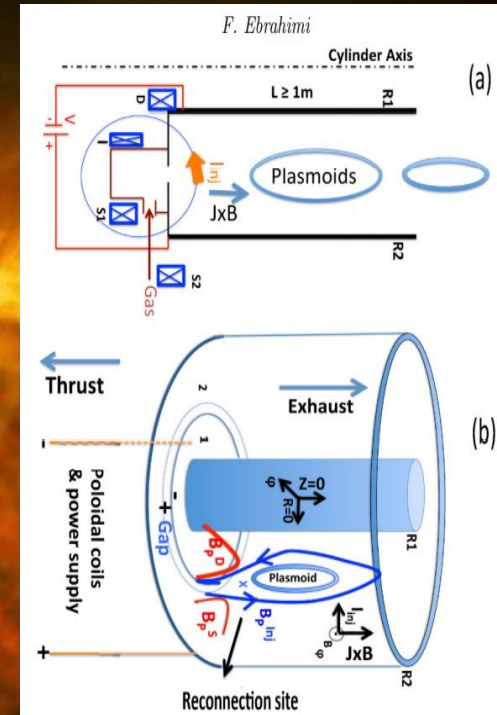
New concept for plasma rocket exploits the mechanism behind solar flares

The magnetic reconnection plasma drive



Princeton plasma physics experiment and simulations show evidence of new acceleration method

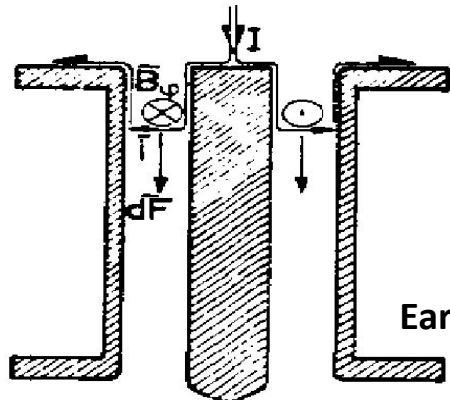
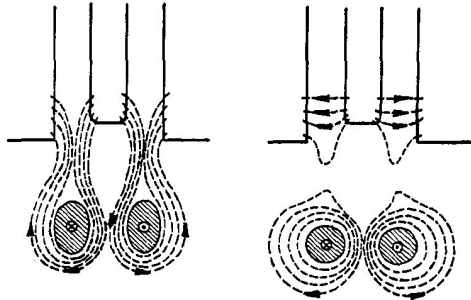
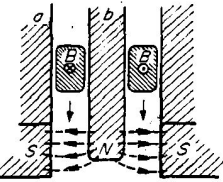
Ebrahimi/Princeton University patent pending USPTO No. 63/085,660
Ebrahimi JPP 2020



Magnetic helicity injection is combined with axisymmetric fast magnetic reconnection



Experiments with plasma rings

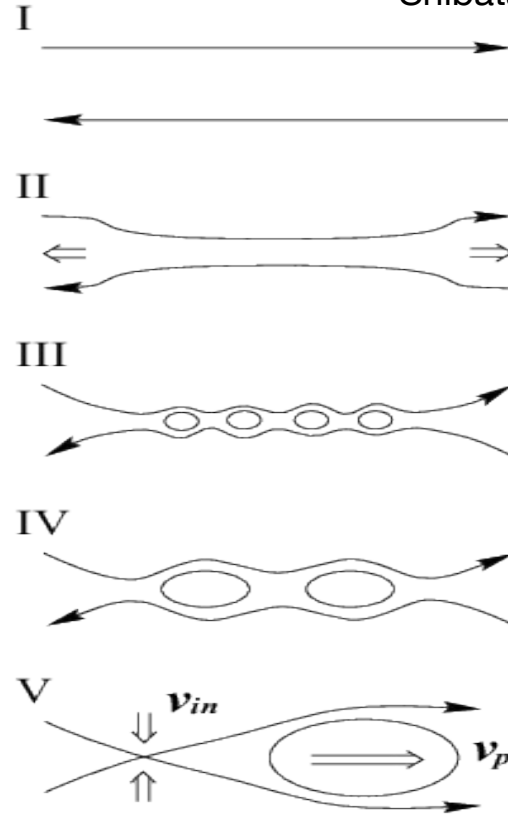


$$d\bar{F} = \bar{i}dv \times \bar{B}_\varphi$$

$$\bar{B}_\varphi = \mu_0 \frac{I}{2\pi r}$$

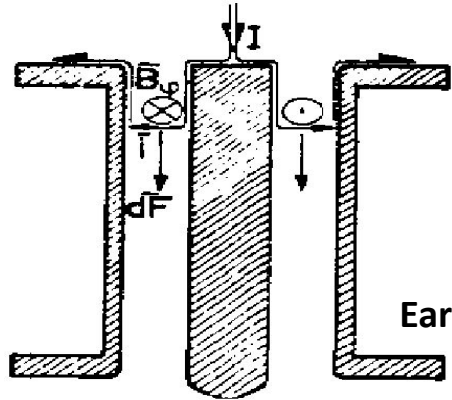
Early experiments of plasma rings

Shibata & Tanuma 2001



Plasmoid reconnection in a current sheet

Magnetic helicity injection is combined with axisymmetric fast magnetic reconnection

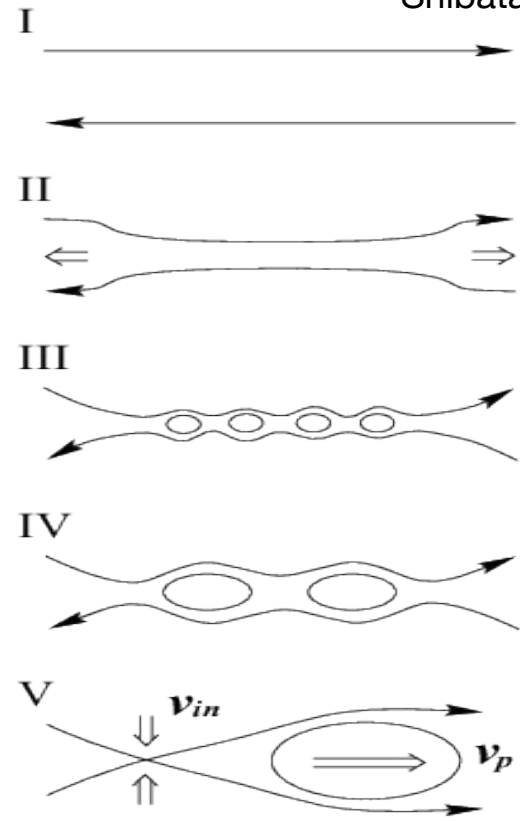


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Plasmoid reconnection in a current sheet



- Solves the linear and nonlinear MHD equations

$$\frac{\partial \mathbf{B}}{\partial t} = -\nabla \times \mathbf{E} + \kappa_{divb} \nabla \nabla \cdot \mathbf{B}$$

$$\mathbf{E} = -\mathbf{V} \times \mathbf{B} + \eta \mathbf{J} + \frac{1}{ne} \mathbf{J} \times \mathbf{B}$$

$$\mathbf{J} = \nabla \times \mathbf{B}$$

$$\frac{\partial n}{\partial t} + \nabla \cdot (n\mathbf{V}) = \nabla \cdot D \nabla n$$

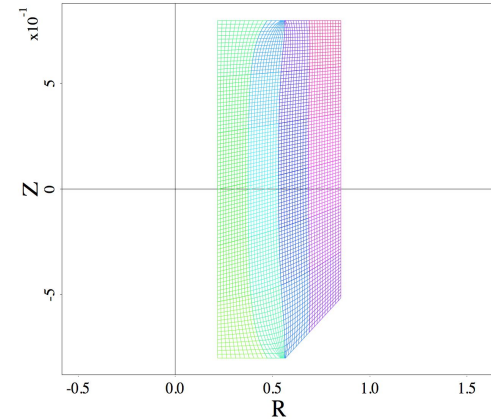
$$\rho \left(\frac{\partial \mathbf{V}}{\partial t} + \mathbf{V} \cdot \nabla \mathbf{V} \right) = \mathbf{J} \times \mathbf{B} - \nabla P - \nabla \cdot \Pi$$

$$\frac{n}{(\Gamma - 1)} \left(\frac{\partial T_\alpha}{\partial t} + \mathbf{V} \cdot \nabla T_\alpha \right) = -\rho_\alpha \nabla \cdot \mathbf{V} - \nabla \cdot \mathbf{q}_\alpha + Q$$

- $\mathbf{q} = -n[(\kappa_{||} - \kappa_{\perp})\hat{b}\hat{b} + \kappa_{\perp}\mathbf{I}] \cdot \nabla T$

- Π is the stress tensor (also includes numerical $\rho \nu \nabla V$)

Finite Element Mesh



- Poloidal grid 45 x 90 fifth/sixth order finite elements, 2-D ($n=0$) and 3-D (up to $n=22$ toroidal modes)

- Voltage is applied across the injector gap (V_{inj}) $E \times B$ normal flows at the gaps

[Bayliss, Sovinec & Redd 2011; Hooper et al. 2012; Ebrahimi et al. 2013; Hooper et al. 2013]

Sweet-Parker current-sheet scaling needs to be incorporated in the classical tearing model



Now \underline{a} (current sheet width) must be replaced with the S-P scaling $\underline{a} \sim L/\sqrt{(S_L)}$

$$\tau_A = \underline{a}/V_A \implies L/(\sqrt{(S_L)}V_A \implies 1/\sqrt{S_L}\tau_A$$

$$S = \underline{a}V_A/\eta \implies L/(\sqrt{(S_L)}V_A/\eta \implies S_L^{1/2}$$

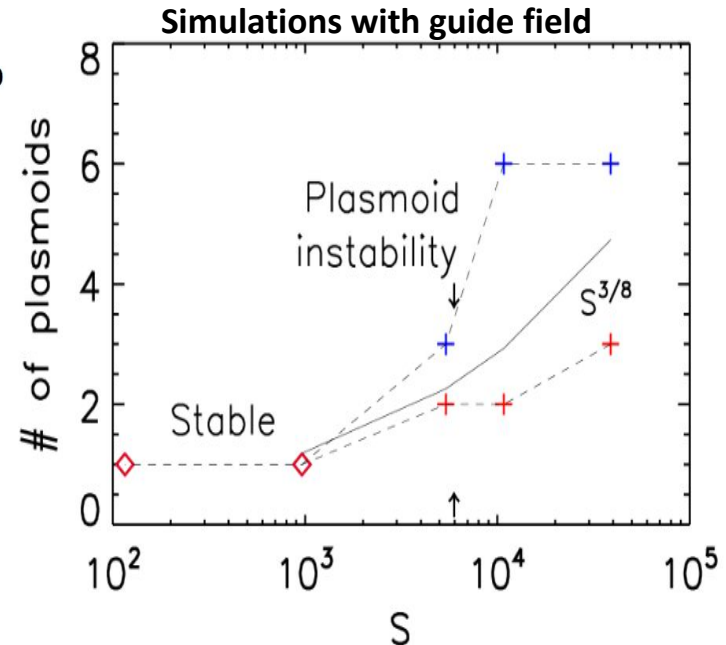
Replacing above into the maximum growth rate eq. gives:

$$\gamma_{\max}\tau_A \sim S_L^{1/4} \quad \text{at} \quad (k_{\max}L \sim S_L^{3/8})$$

(Tajima & Shibata 1997 Plasma Astrophysics, Loureiro et al. 2007)
& Bhattacharjee et al. 2009

Here : $\tau_A = L/V_A$ and $S_L = LV_A/\eta$
(L is the length of the current sheet)

Now the growth rate has a positive exponent S scaling



Number of plasmoids is an increasing function of S. Blue: small sized transient plasmoids Red: large scale and persistent plasmoids.

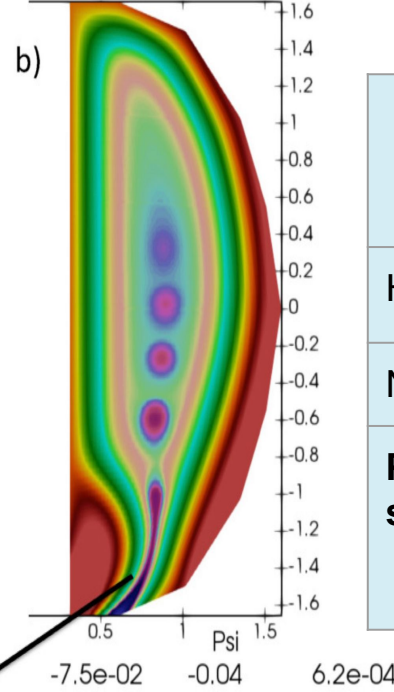
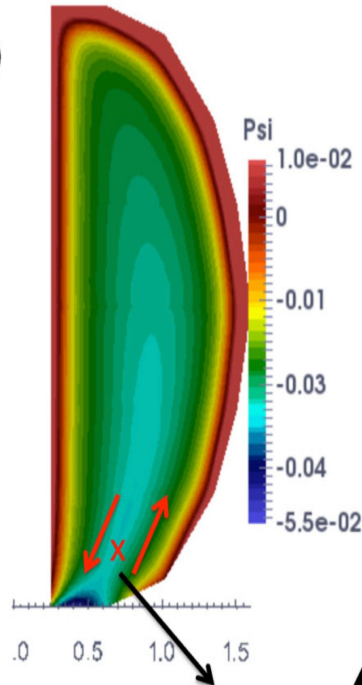
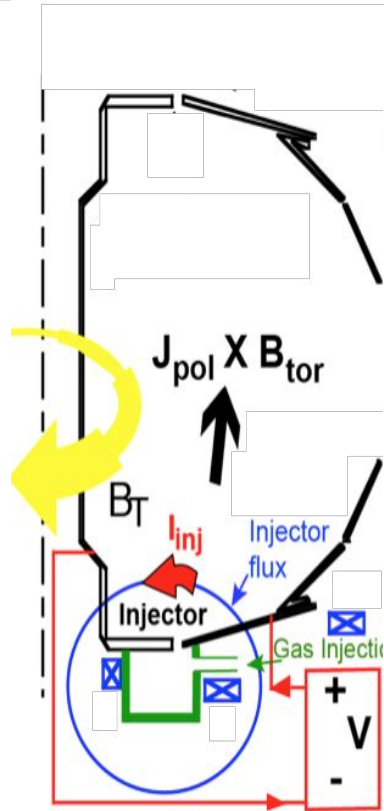
Ebrahimi&Raman PRL 2015

Experiments and simulations in NSTX have proven the Transient Coaxial Helicity Injection (CHI) as a primary and scalable candidate for solenoid-free startup in STs



During decay ($V_{inj} = 0$)

During injection (V_{inj} on)



Reconnection sites

	BT [T]	Injected flux (Wb)	Startup current (MA)
HIT-II	0.4	0.03	0.1
NSTX	0.6	0.06	0.2
Projected simulation	4	0.3	1

(NSTX plasma volume is $\sim 30x$ HIT-II)

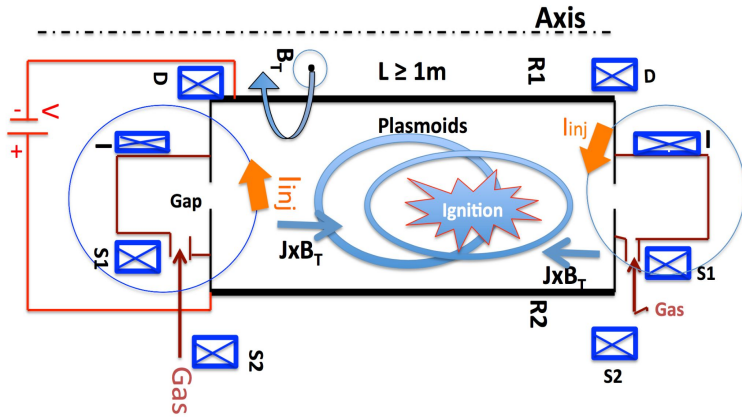
[Experiments: Raman et al. PRL 2003, 2006]

[Simulations: Ebrahimi PoP 2019, 2016, Ebrahimi et al. PoP 2013, Ebrahimi&Raman PRL 2105, NF 2016]

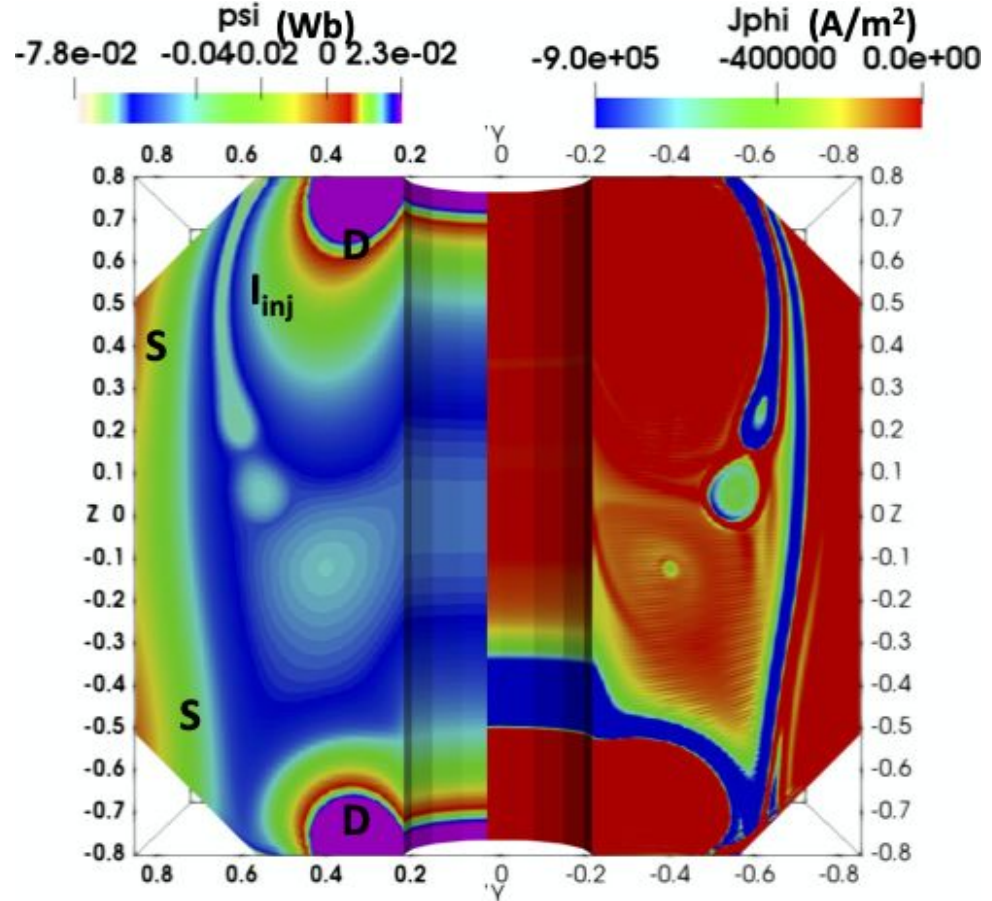
Novel fusion concept using plasmoid reconnection (a double null configuration)



Plasmoid reconnection fusion concept.
Princeton University (USPTO No. 63/085,660)
Ebrahimi



Continuous merging using helicity injection plasmoids. All the fueling and heating is done by plasmoid reconnection.



“Nuclear Fusion Inspires New Rocket Plasma Thruster Design”

2015 -2016



Proof of Principle (foundation)

State-of the art simulations
+NSTX experiment

PRL 114, 205003 (2015)

PHYSICAL REVIEW LETTERS

week ending
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Plasmoids Formation During Simulations of Coaxial Helicity Injection in the National Spherical Torus Experiment

F. Ebrahimi¹ and R. Raman²

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Ebrahimi&Raman PRL 2015

Ebrahimi &Raman NF 2016, FE PoP2016

➤ **Fast magnetic reconnection creates non-inductive current for fusion**

2017



Propulsion idea

Original invention disclosure filed with Princeton University on June 27, 2017

2018-2019

Favorable current scaling: for high fusion current (possible for ohmic ignition)

Ebrahimi PoP 2019

2020-2021

Provisional patent filing

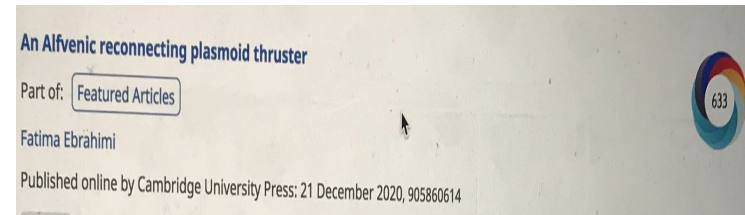
September 29 2021

Submitted peer-review paper Feb 6 2020

Ebrahimi JPP 2020

DOE internal funding - Team Engineering design/2022 -

2022-2023

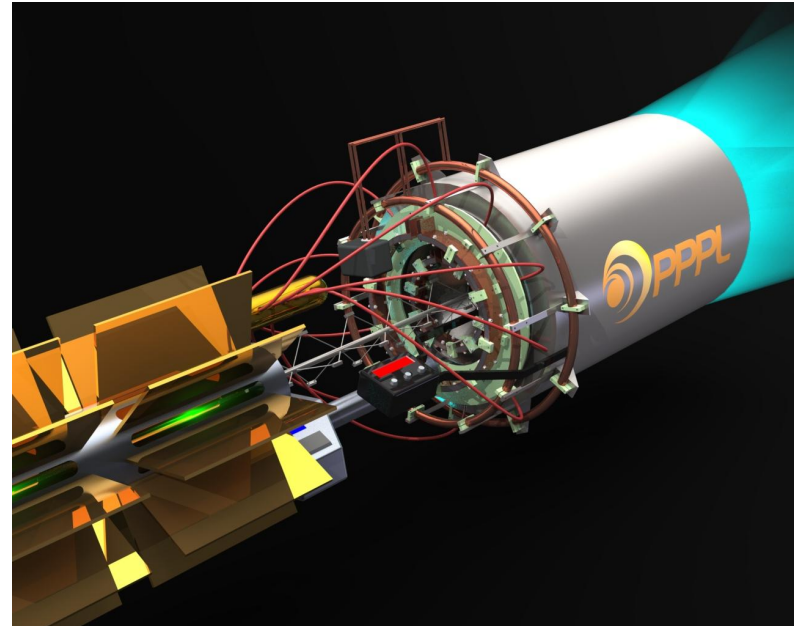


➤ **New propulsion concept based on fast magnetic reconnection proposed**

An Alfvénic Reconnecting Plasmoid Thruster

Princeton University (USPTO No. 63/085,660)

- We are in the design/build stage of a magnetic reconnection thruster at PPPL using a 3-year internal lab funds.



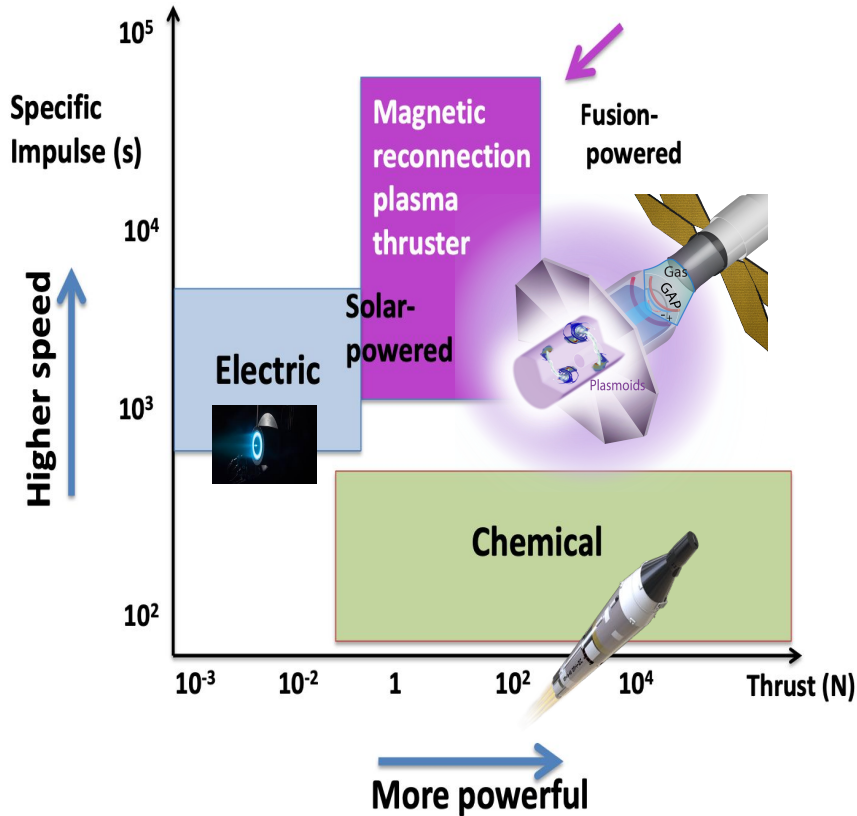
An Alfvénic reconnecting plasmoid thruster

Part of: [Featured Articles](#)

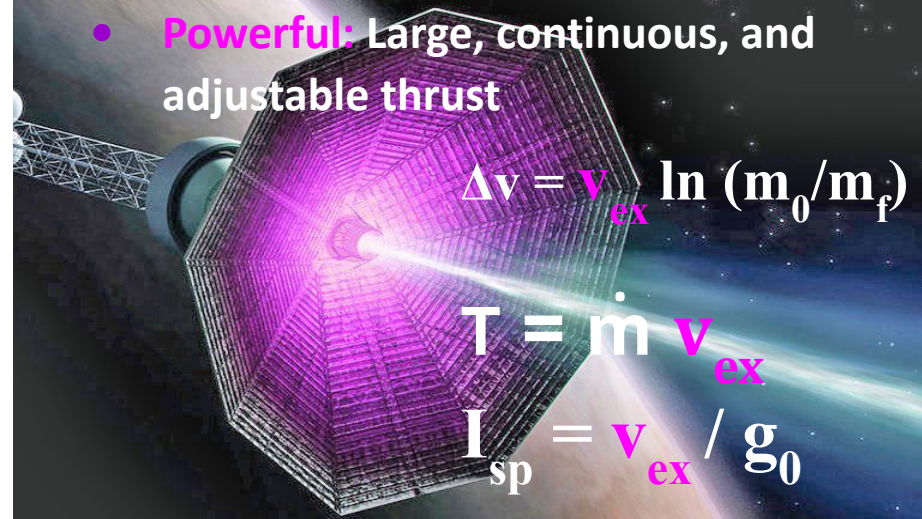
Fatima Ebrahimi

Published online by Cambridge University Press: 21 December 2020, 905860614





- **Easier refueling:** Flexible propellant
- **Efficient:** Accelerate plasma ions to high velocity
- **Powerful:** Large, continuous, and adjustable thrust

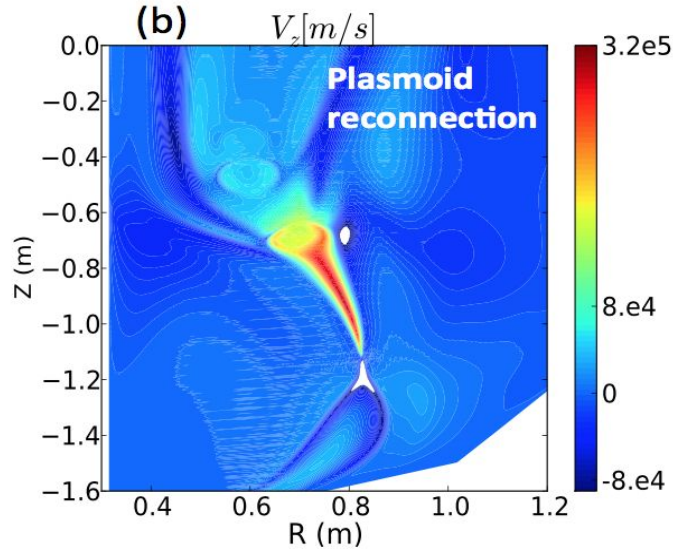
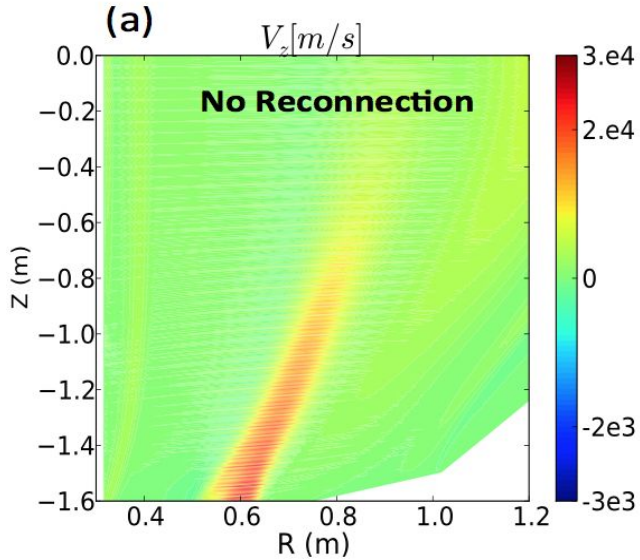


$$\Delta v = v_{ex} \ln(m_0/m_f)$$

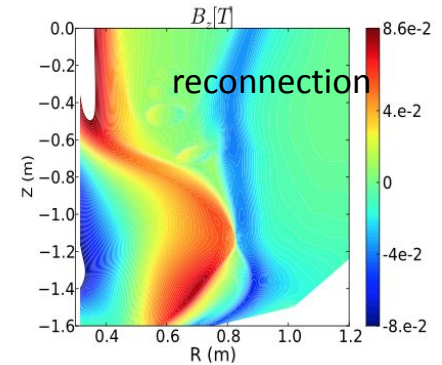
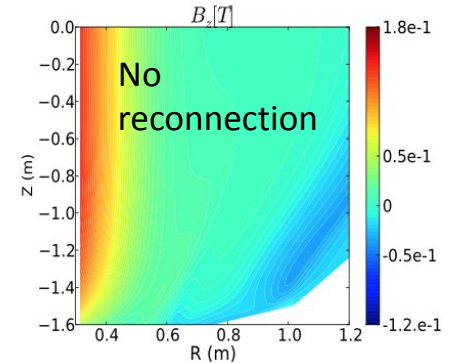
$$T = \dot{m} v_{ex}$$

$$I_{sp} = v_{ex} / g_0$$

Magnetic reconnection is essential for high exhaust velocity

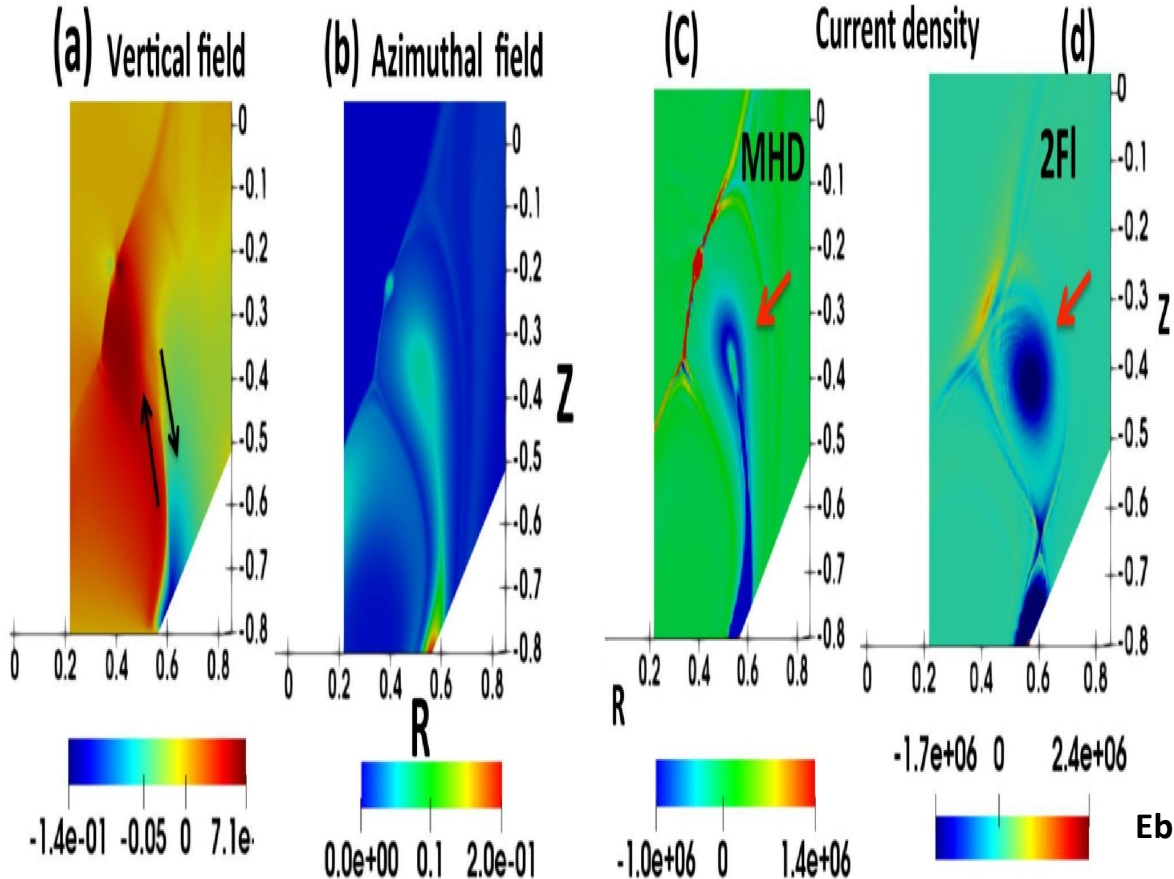


Vertical fields



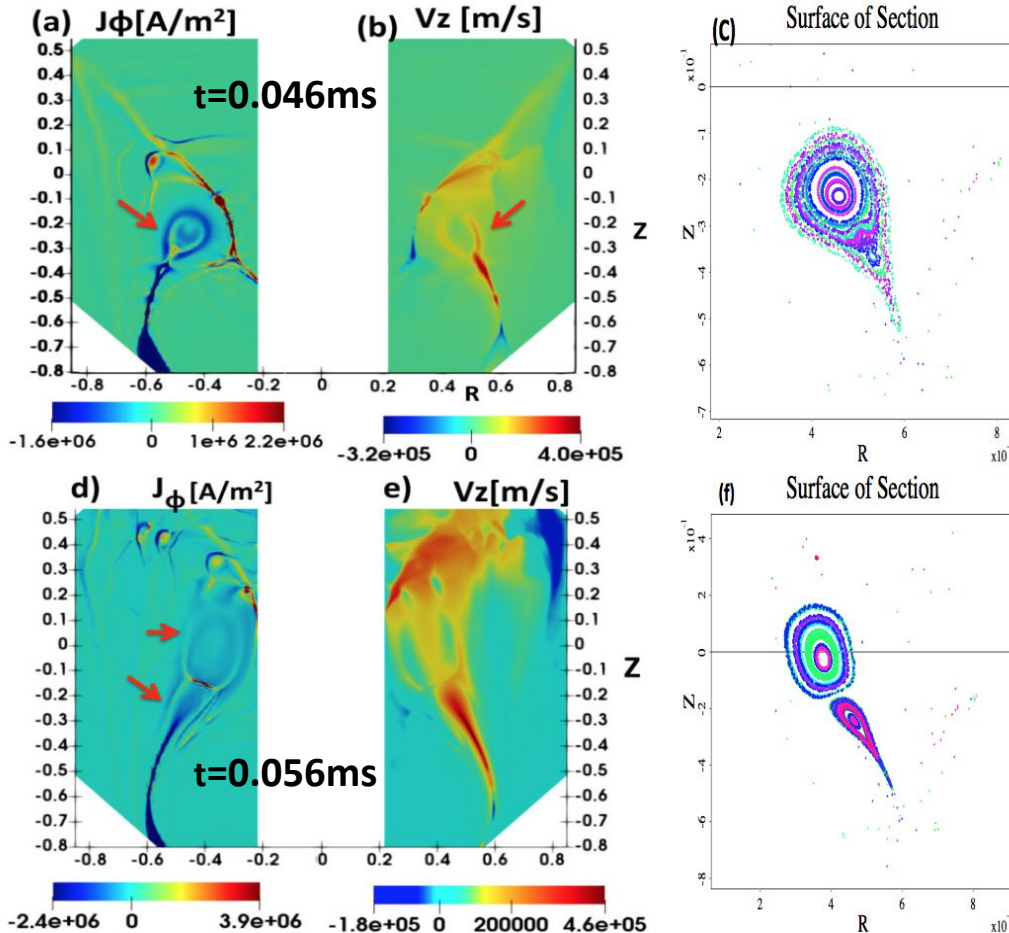
- Outflow velocities obtained in the simulations
- Magnetic configuration has essential role in inducing reconnection in this thruster

The plasmoid instability is triggered



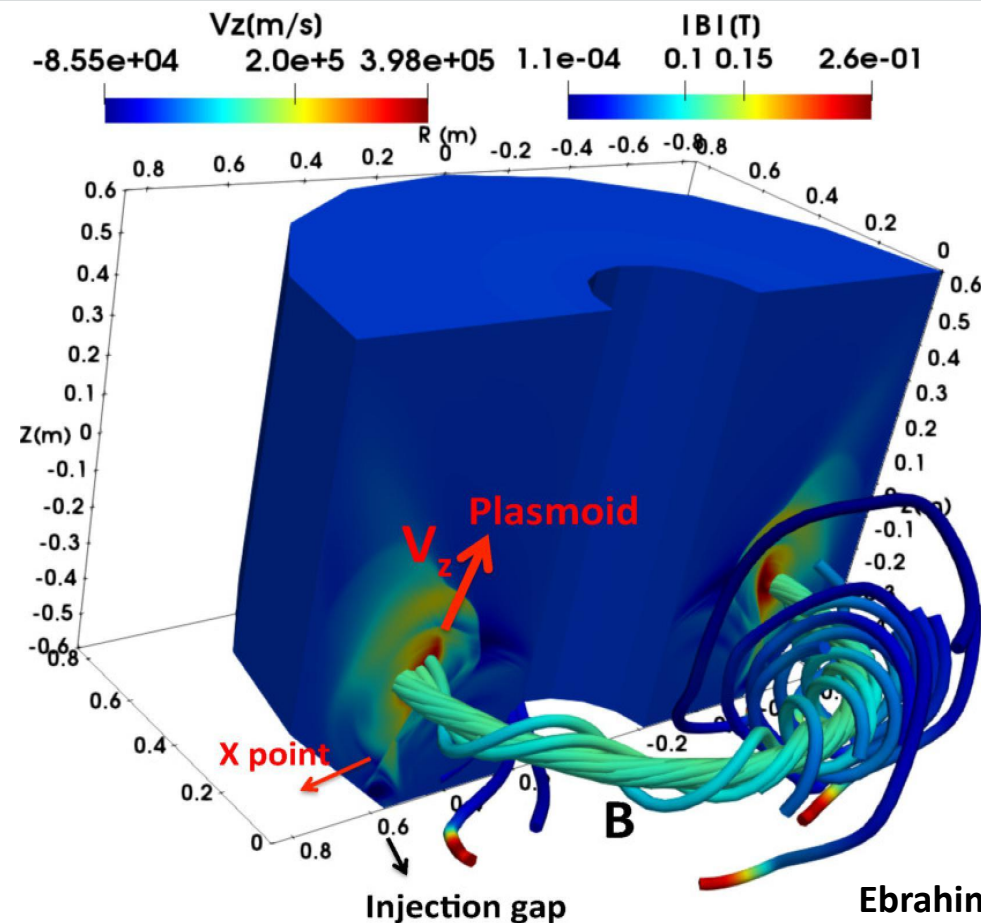
- The static fields B_p^D and B_p^S radially pinch the injector field around the injector gap to form a primary exhaust reconnecting current sheet
- The plasmoid instability is here triggered at local Lundquist number $S \sim 12\,000$ (based on $B_z \sim 500$ G, $L = 0.5$ m and $\eta = 16$ m² s⁻¹).

Cyclic ejection of large-scale axisymmetric plasmoids is observed

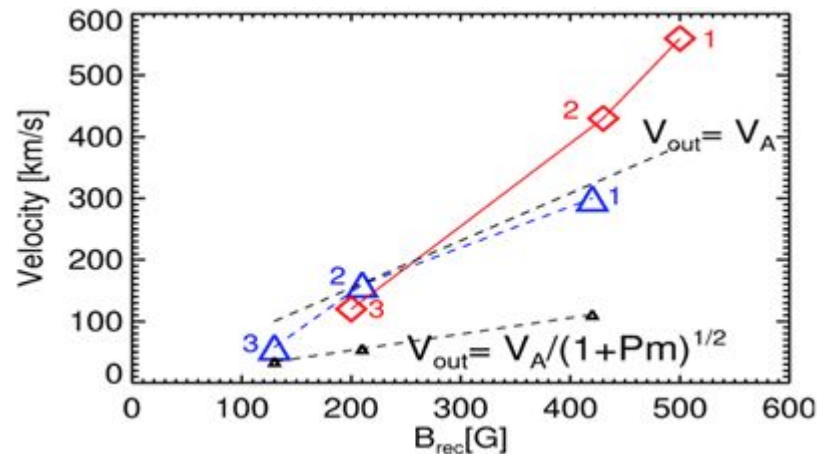


- As the first plasmoid is already ejected, the open field lines start to close again and a large-volume closed field line in the form of a second plasmoid is formed and departs the device with a high outflow velocity of about 460 km s⁻¹.
- The maximum outflows are both along the exhaust current sheets in the form of outflow jets, as well as along the plasmoids.
- Cyclic ejection of large-scale axisymmetric plasmoids is around 10 μs (4–5 Alfvén transit times).

Momentum-carrying plasmoids are formed during three-dimensional global MHD simulations



Exhaust velocity scales with magnetic field



- $V_{exhaust} \sim V_{A(pol)} = B_{in(z)} / \sqrt{(\mu_0 \rho)}$.
- $F = \rho V_A^2 A$ <--- **Thrust**
- ISP $\sim 2,000 - 50,000$ s
- Thrust $\sim 1 - 100$ N (Power $0.1 - 10$ MW)
- For the high-power case that corresponds to about $5-10$ mN/kW.

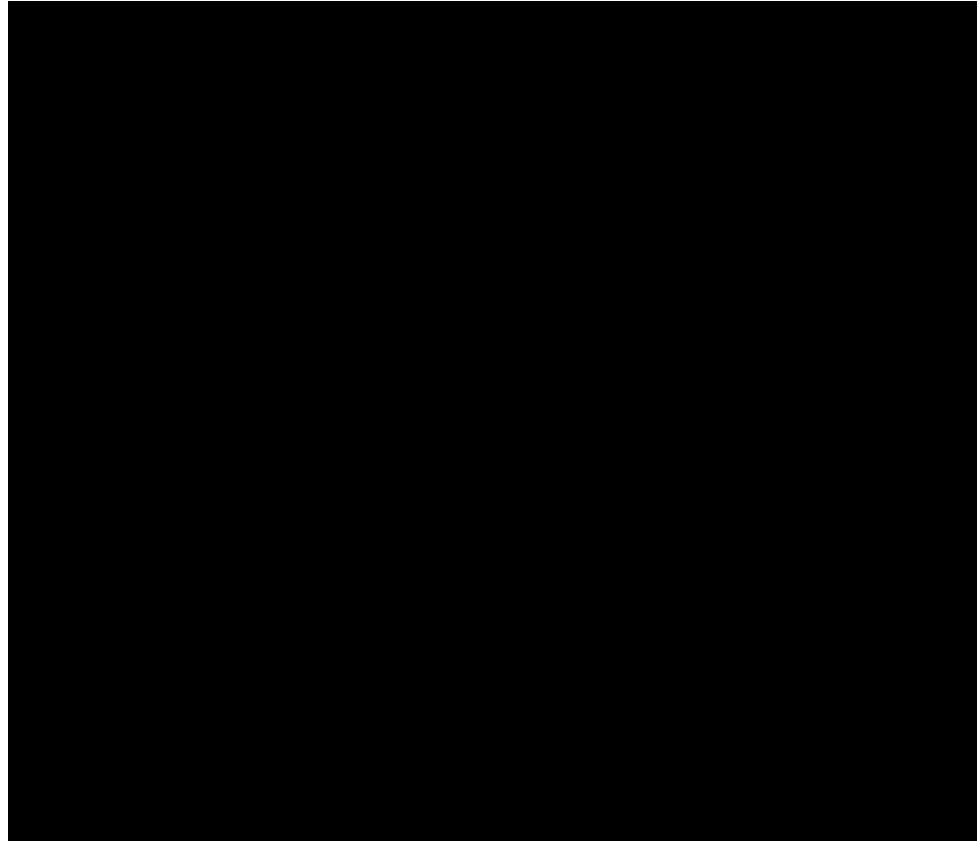
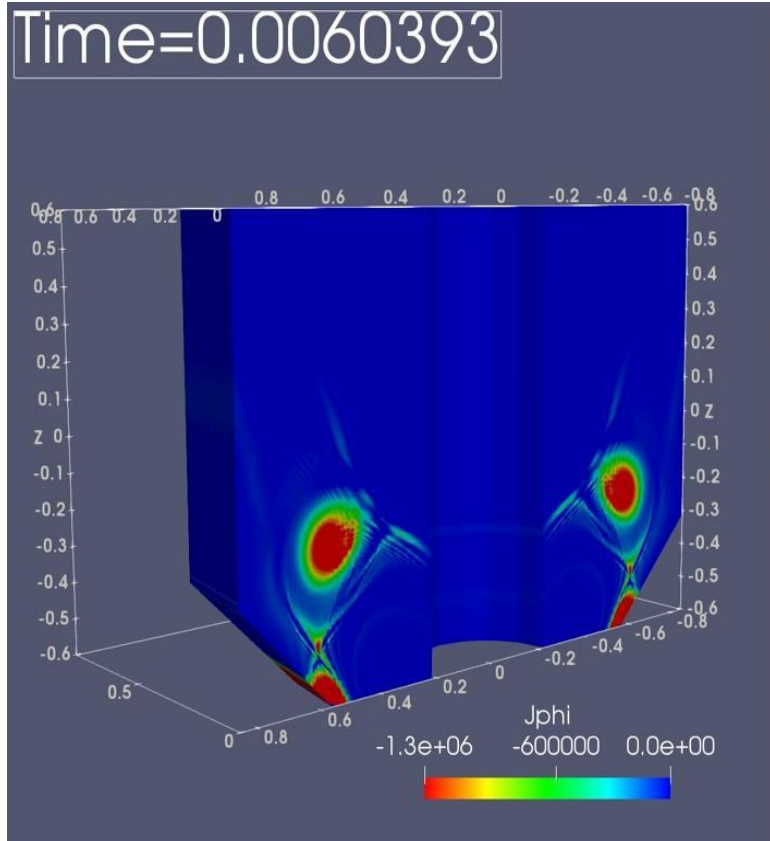
Subsequent ejection of a large plasmoid during Hall reconnection with large exhaust velocity



Current density

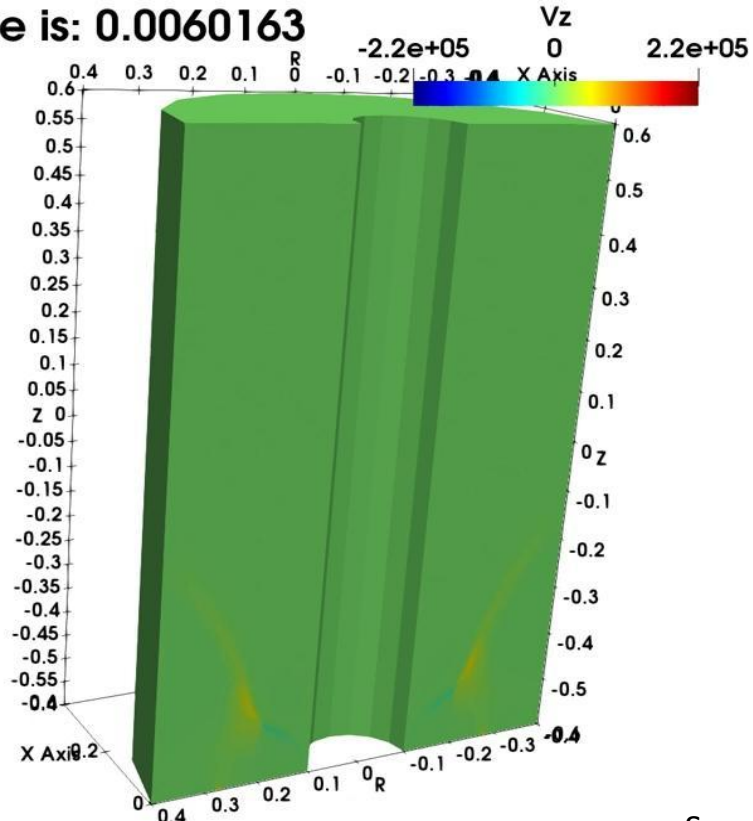
Exhaust velocity

Time=0.0060393

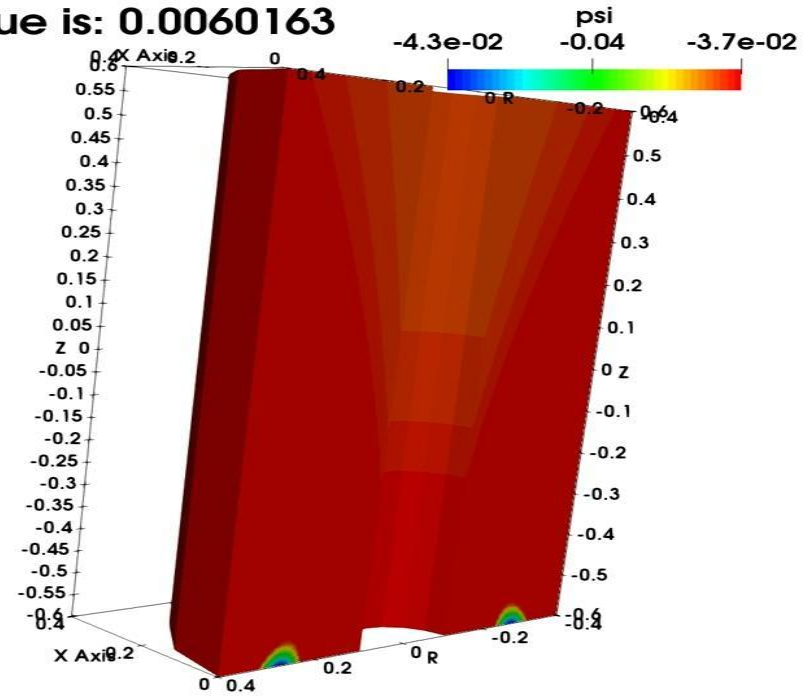


MHD simulations for a small-scale device

Value is: 0.0060163



Value is: 0.0060163

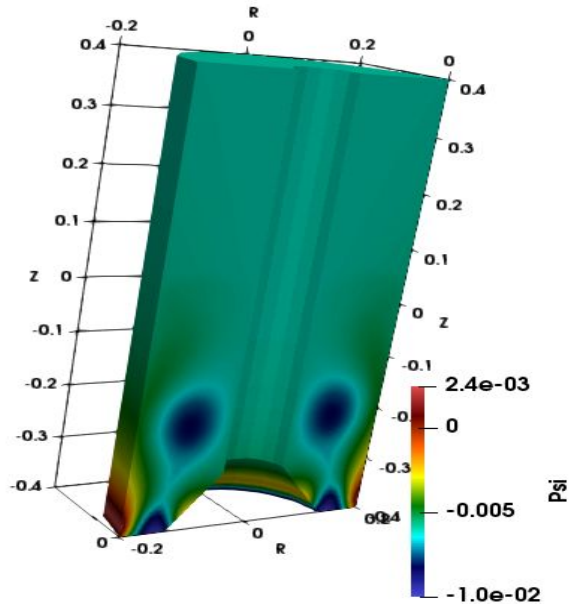


- Several magnetic configurations for smaller scale device performed.

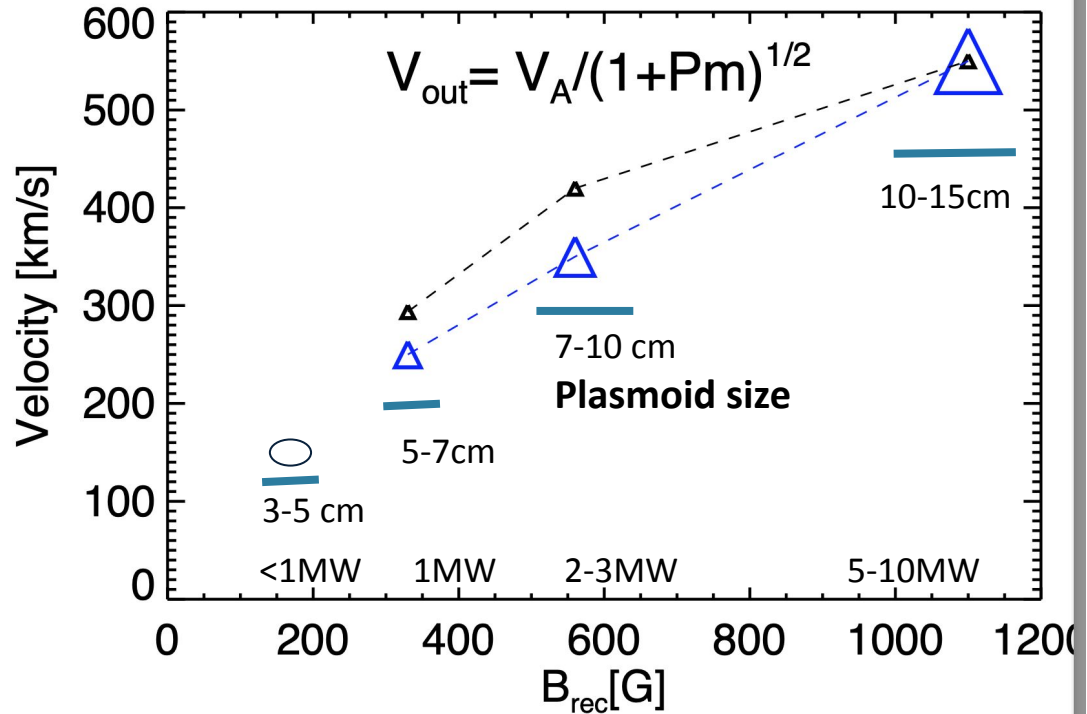
Favorable magnetic field scaling also obtained



Time: 0.00601514



- The size of the large plasmoid is for the high power case.



- Plasmoid size as well as **average** speed increases (up to ISP of 50000 s) with the reconnecting field. No guide field here.

PMRT System Design (building at PPPL)



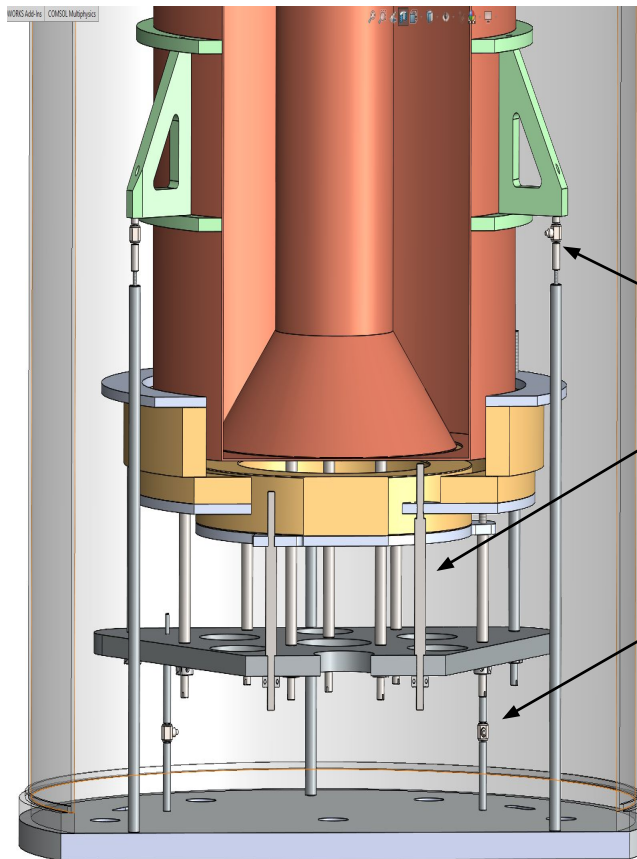
Simple structure, low-cost in manufacturing/assembling

- **Vacuum Vessel:**
40" × 70" 5-mm Stainless steel

- **Plasma Chamber:**
ID: 0.1-m – OD: 0.3-m – Length: 1.2-m
Copper (Glidcop) with tungsten coating

- **Thrust measurement:**

- Can be changed to horizontal arrangement by adding supporting structure



Force sensor integrated to supporting structure

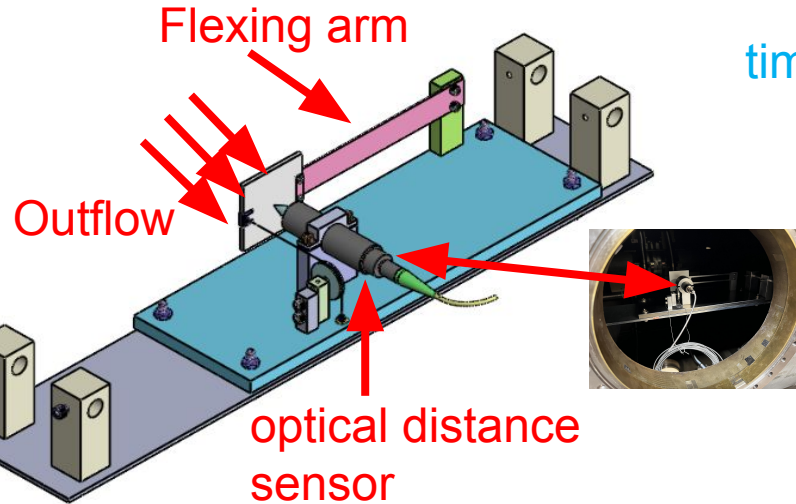
All coil locations adjustable by adjustment screws

1. Multiple sensors for measuring asymmetrical force

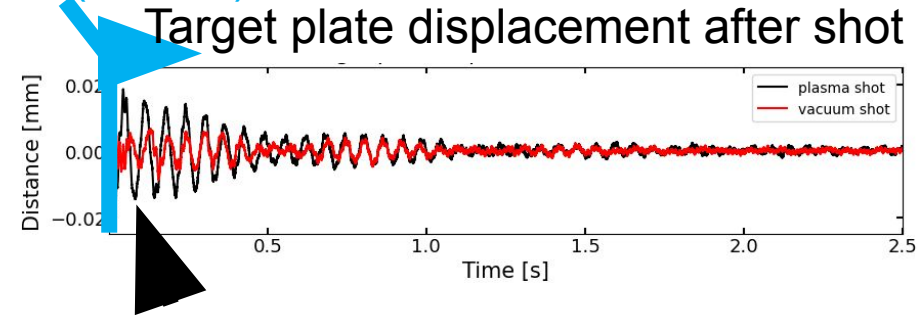
- No electrodes for lower electrode erosion
- Generate asymmetry using outflow direction \mathbf{B} pressure gradient
- Initial displacement measurement data (for thrust) on MRX is promising

Kush Maheshwari+ poster

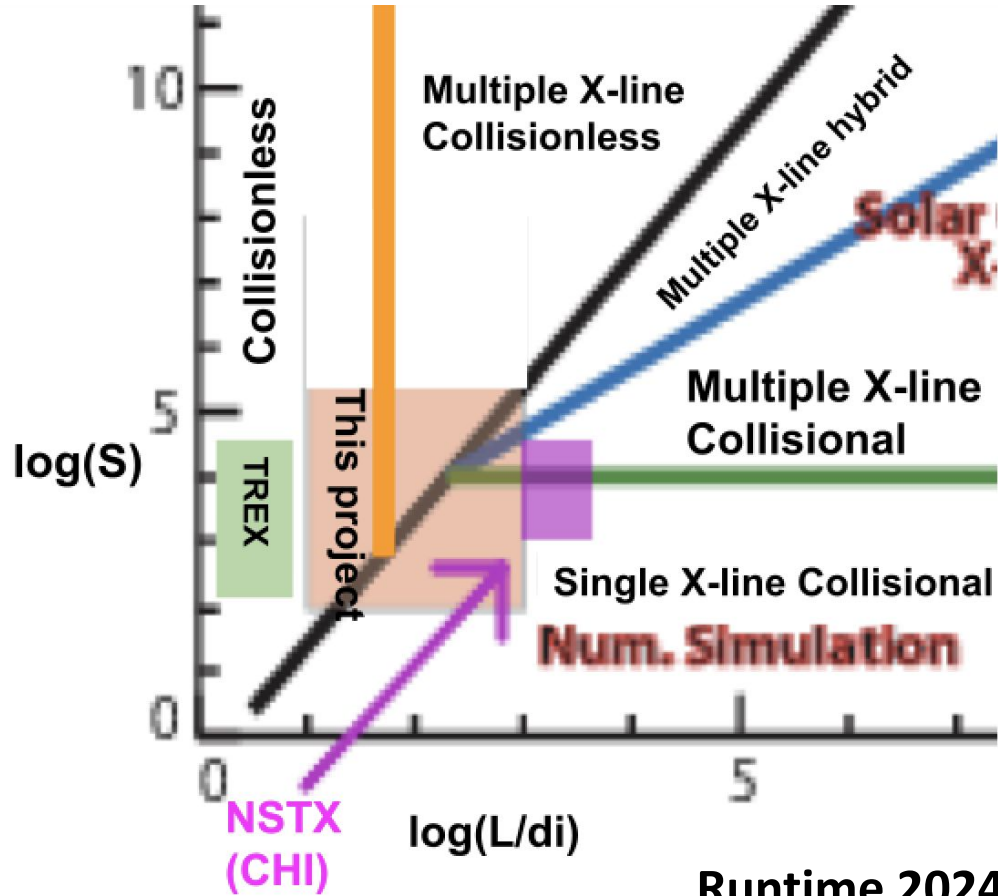
Displacement/thrust measurement apparatus



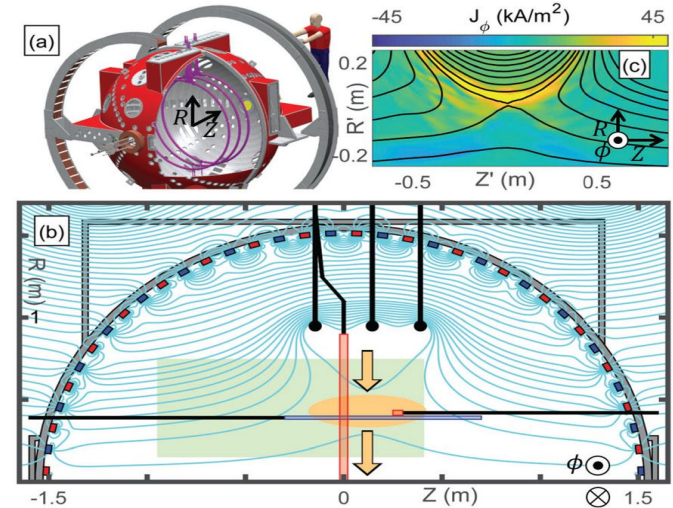
time of shot (< 1 ms)

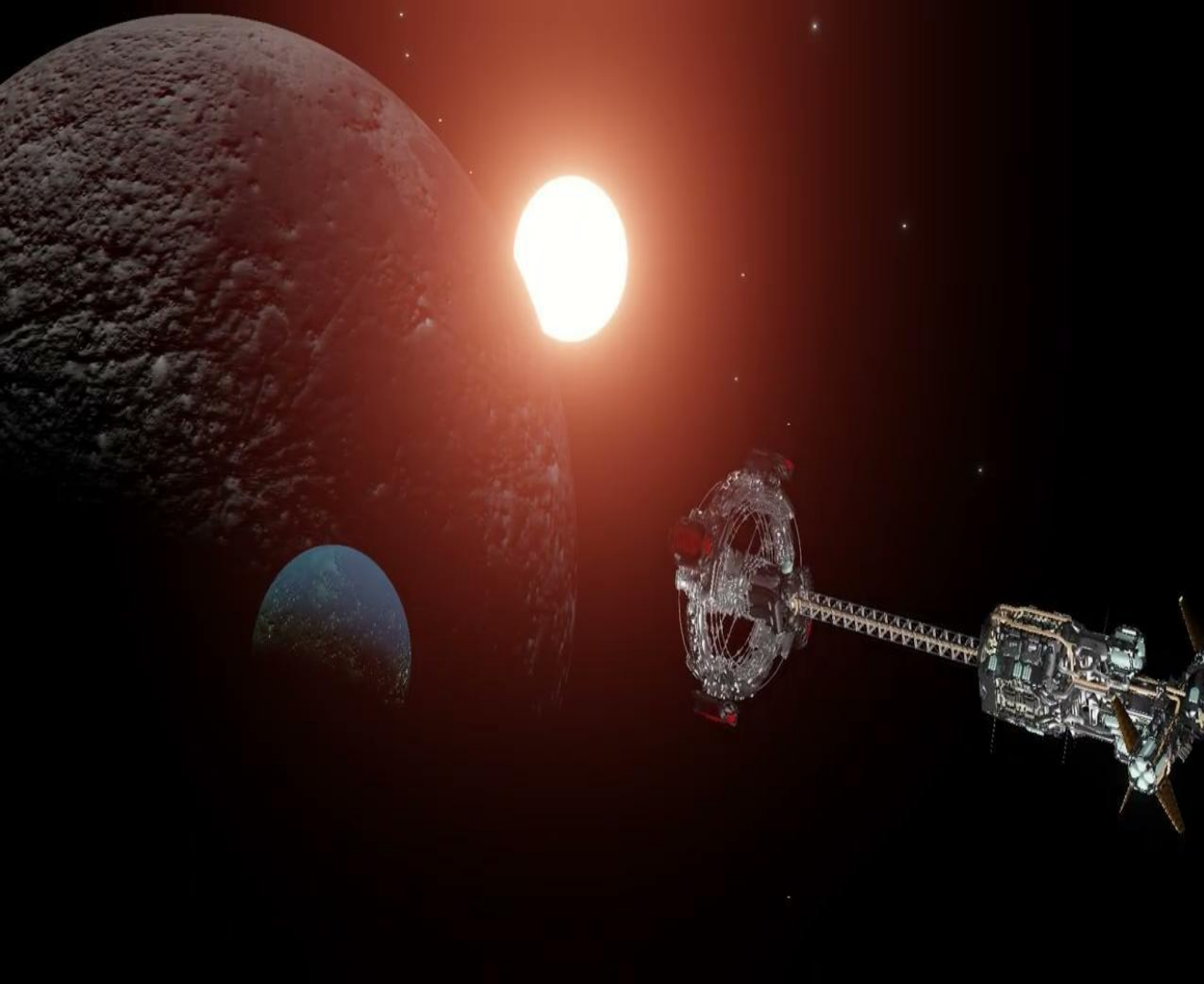


Plasma shot $>2x$ amplitude compared to vacuum shot



- Orange box: desired operational regime, approaching the MHD collisional limit ions contribute in the reconnection process for propulsion $\lambda \sim > 40$**





oid reconnection thruster concept
include a unique fusion concept.
ity (USPTO No. [63/085,660](#))



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- **Understand and predict** 3-D MHD simulations have shown that system-size plasmoid formation can produce large plasma startup current in spherical tokamaks and a large-fraction conversion of injected open flux to closed flux.
- **Explore** Continuous plasmoid injection due to spontaneous reconnection could be the key for steady-state current-drive. New magnetic configuration is being investigated.

A reconnecting plasmoid thruster is highly scalable

- **Exploits magnetic reconnection to convert magnetic energy to kinetic energy with high exhaust velocity**
 - Experiments at NSTX and simulations using state-of-the-art 3D computational models running on NERSC supercomputers provide a firm scientific foundation for the propulsion concept